DATE PREPARED:		ISS PAYLOAD OFFICE				PAGE 1 OF 5			
June 13, 2007		IRN/EXCEPTION FORM							
Doc. No., Rev. & Title: SSP 57213, Initial Release				PIRN No:					
Alpha Magnetic Spectrometer (AMS) Interd Document			erface Control	ace Control 57213-NA-003		011			
TIT	LE:			1					
	AMS (Alpha Magn	etic Spectr	ometer) – PVG	F Cable T	ie Down	Exceedar	nce		
Originator:			PIRN Type:	PIRN Type: Standard PIRN		FAX Approval Signatures to this Number: 314-777-2866			
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Agency Tracking No.: 57213-0007			SYSTEM/ELF	SYSTEM/ELEMENT AFFECTED & STAGE EFFECTIVITY:					
			AMS – Launo	AMS – Launch through End Of Life					
	ASON FOR CHANGE OR REQUIREME								
Αľ	MS exceeds PVGF cable tie-d	lown requir	ement defined in	SSP-57003	8, Paragra	phs 3.7.3.	3.4-B.		
PARAGRAPHS, FIGURES, TABLES AFFECTED (For PIRN use only)									
Page Paragraph(s) Figur		ures(s)	Table(s)			<u>R</u> <u>A</u>	<u>D</u>		
3-75 3.7.3.3.4-B N/A		\	N/A						
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	AFFECTED INTERFACING PARTIES ASSOCIATION DUTIES OF CONTRACTOR OF CONTRA								
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	S&MA	F	ER3						
	AMS		OM7						

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SSP 57003 Requirement:

Requirement Number: 3.7.3.3.4-B

Requirement Title: ATTACHED PAYLOAD ELECTRICAL INTERFACE

HARDWARE

B. The PVGF cable harness shall be secured to the payload within 6.0 inches of the cable harness exit bracket. The user shall be responsible for routing and securing the harness to ensure it is outside of the SSRMS and EVA clearance zone as specified in NSTS–21000–IDD–ISS.

Proposed AMS Payload Exception:

Requirement Number: 3.7.3.3.4-B

Requirement Title: ATTACHED PAYLOAD ELECTRICAL INTERFACE

HARDWARE

B. The PVGF cable harness shall be secured to the payload within 6.0 inches of the cable harness exit bracket. The user shall be responsible for routing and securing the harness to ensure it is outside of the SSRMS and EVA clearance zone as specified in NSTS-21000–IDD-ISS, except for the AMS payload that will secure the cable within ~15 inches as there are no other tie-down points as can be seen in Figure 3.7.3.3.4-AMS-01: AMS PVGF Cable Tie-Down Violation.

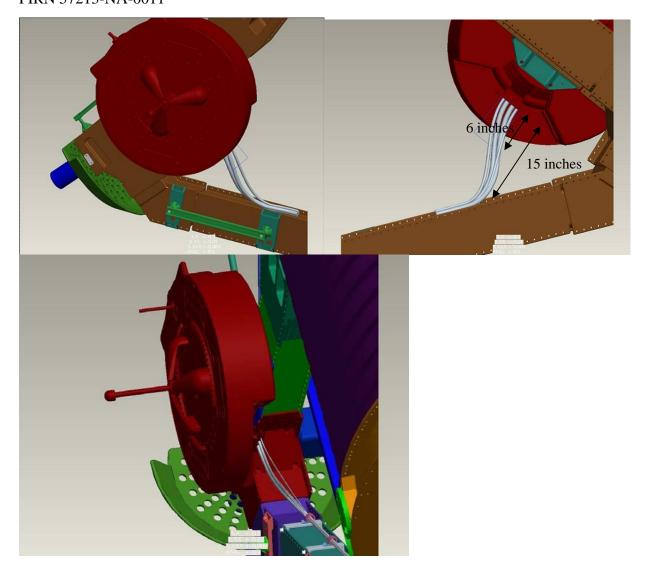


Figure 3.7.3.3.4-AMS-01: AMS PVGF Cable Tie-Down Violation

Rationale:

As shown in the figure, there are no tie-down points available at the required 6 inch location. The first tie-down location is approximately 15" from the PVGF cable departure point. There are no alternative PVGF mounting locations. Rotating the PVGF does not work with the current design. Standard practice in the aerospace industry calls for cable tie-downs every 18 inches, so we believe that this should be acceptable.

PEI Analysis:

A vibration issue was identified for this violation. ER3 requested an analysis from MacDonald Dettwiler Space and Advanced Robotics Limited (MDA). MDA's report SRMS.123638 Rev A determined that the PVGF cable exit clamp will have no structural issues. Based on this report, ER3 provided concurrence via e-mail (attached) on January 16, 2009. PEI agrees with MDA's analysis results.

Per Crew Consensus Report (CCR): AMS NBL 11.12.02 (Attached), EVA access to the PVGF contingency release bolt is acceptable. Boeing EVA provided PEI with Figure 3.7.3.3.4-AMS-02: PVGF EVA Bolt and SSRMS LEE EVA Bolt (shown below) to demonstrate that the AMS PVGF will not impact a crewmember access to both of these EVA bolts. PEI concurs with CCR AMS NBL 11.12.02 and with Boeing EVA Figure 3.7.3.3.4-AMS-02 which demonstrate that the PVGF Cable tie-down requirement violation does not impact EVA.

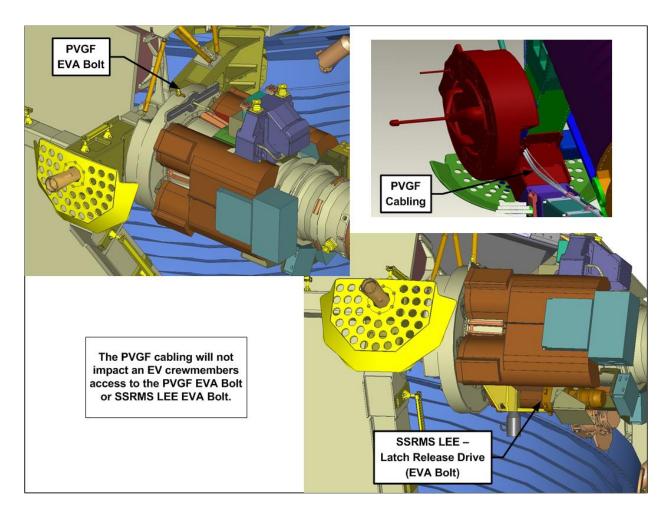


Figure 3.7.3.3.4-AMS-02: PVGF EVA Bolt and SSRMS LEE EVA Bolt

PCB Disposition:

TBD

PIRN 57213-NA-0011 PEI Recommendation: Approve as written. Operational Constraints: (As Needed) None identified by PEI Operational Constraints Rationale: (As Needed) None identified by PEI PTR Recommendation: TBD